

Plasma plume structure of the Alternative Low Power Hybrid Ion Engine (alphie)

UPM PlasmaLab

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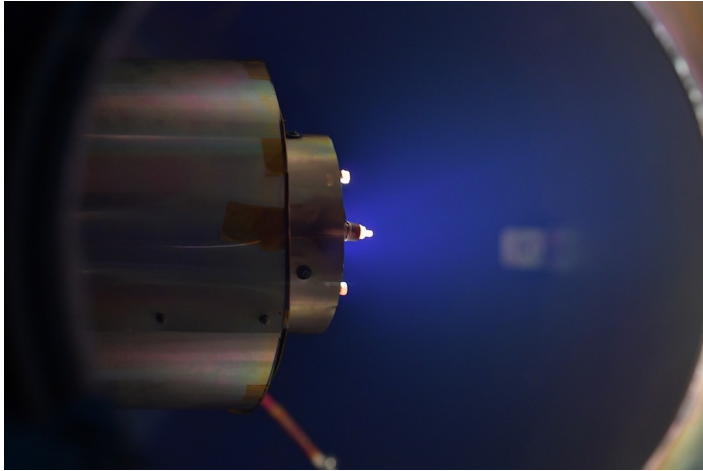


Fig 1. The *alphie* in steady state operation.

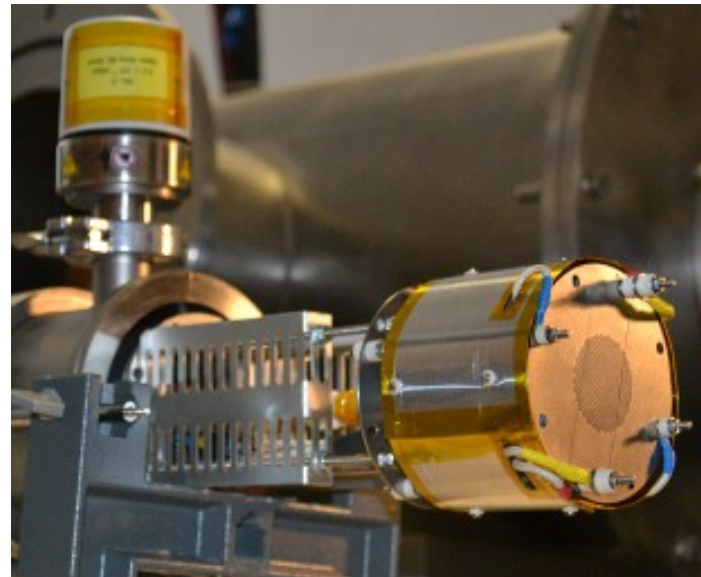


Fig 2. The *alphie* in its support.

- The ALPHIE design is a new technology of a plasma accelerator for satellite propulsion in space.
- This small 10 X 15 cm plasma thruster operates with less than 300 W electric power consumption.
- It is intended for small and medium sized satellites (roughly over 100 Kg) where most commercial propulsive systems are nowadays difficult to implement.
- Four prototypes have already been tested in the laboratory.
- *This technology is free from ITAR restrictions* and two patents have been granted in 2019:

European Patent Office: Patent EP 3369294B1

US Patent and Trademark office: Patent US 10172227

Envisaged applications

- Station keeping
- Orbital drag compensation in LEO/MEO
- End-of-life disposal of satellites
- Flight formation

Operation and characteristics

- Operates with only **3 DC power supplies** and only two are employed in normal operation. Simple PPU design.
- Easy direct electric connection with solar panels.
- Only **one cathode** is employed as electron source for both plasma production and ion beam neutralization. This makes an **important difference with conventional gridded ion engines**.
- Testing new cathode technologies

Magnitude	Value	Commentary
Weight	1.2 kg	Without PPU
Dimensions	10 x 15 cm	Diameter x length
Propellant gas	Ar, Xe	Kr in the future
Gas flow rate	1-2 sccm	
Power consumption	200-325 W	
Thrust	0.8-3.5 mN	Ar, throttleable
Specific impulse	13900-20000 s	
Thrust-to-power ratio	4-11 mN/kW	

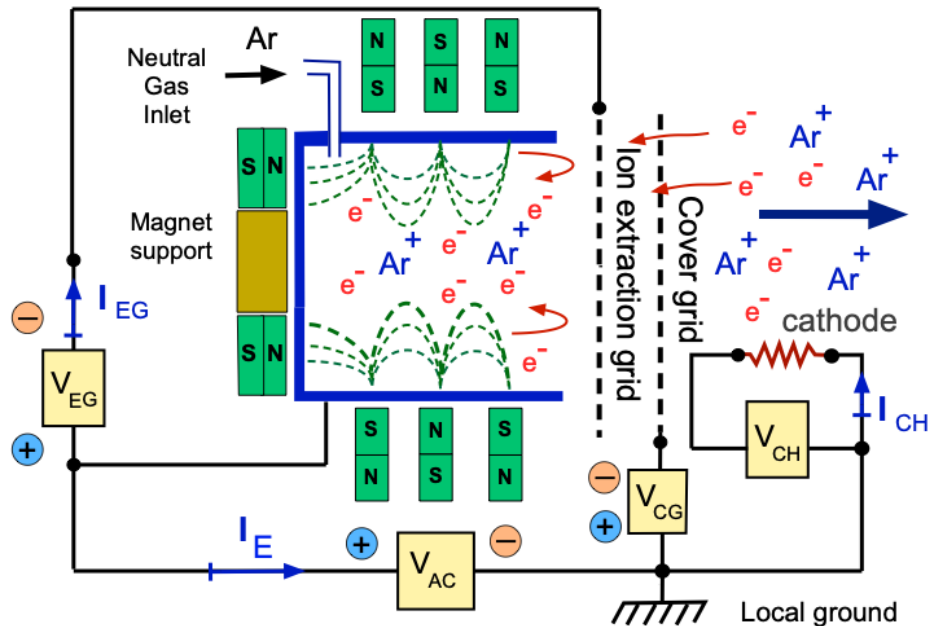


Fig 3. Electrical scheme of the *alphie*.

- Grids are parallel and made of a drilled stainless steel plate and are essential for plasma beam collimation.
- Electrons are trapped by the strong magnetic field.
- Both **ions and electrons counterflow through the two-grids system**.
- Strong interaction between accelerated electrons and neutrals.
- The electric potentials are always below kV range.

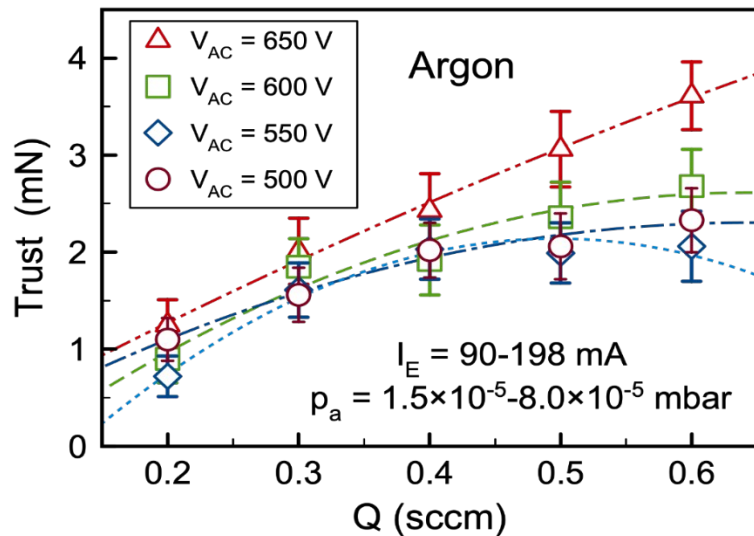


Fig 4. Thrust provided by the *alphie* as a function of mass flow rate (Q) and acceleration voltage (V_{ac}).

- *Variable throttle operation* (see Fig 4) controlled by the acceleration voltage V_{acc} and the mass flow rate Q .
- Future characterization campaign involving Xenon.
- *Alphie* generates two populations of ions for high voltages: Fast and Slow.
- Fast group velocities between 30-60 km/s.
- Slow peak follows an isometric expansion (thermal group). Fast group more collimated.
- For a constant mass flow, two parameters characterize the operation of *alphie*: the electron current I_b , and the acceleration voltage V_{acc} .

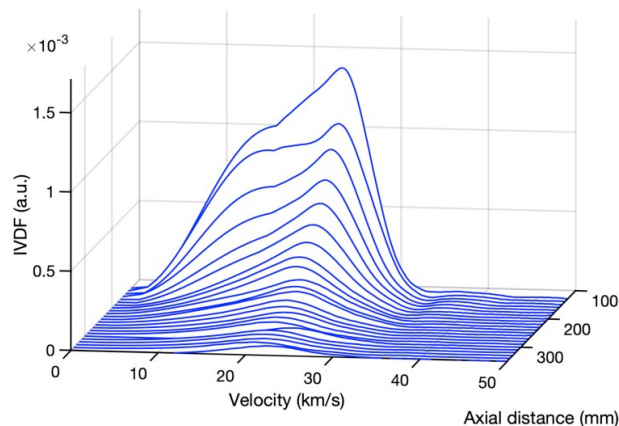


Fig 5. IVDF measured using a retarded field energy analyzer (RFEA) for a low (450V) acceleration voltage.

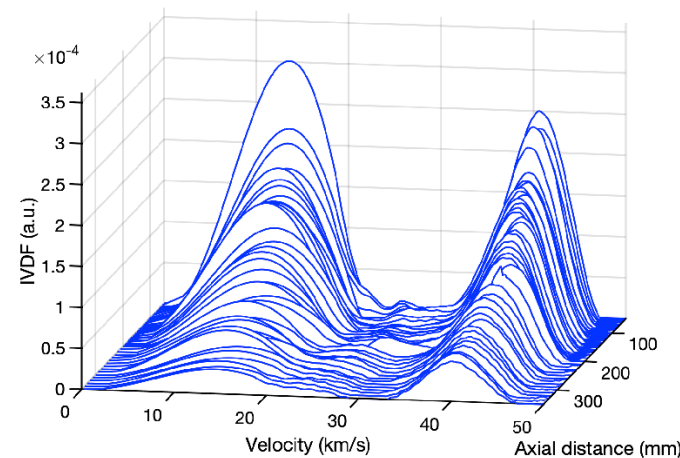


Fig 6. IVDF measured using a retarded field energy analyzer (RFEA) for a high (550V) acceleration voltage.

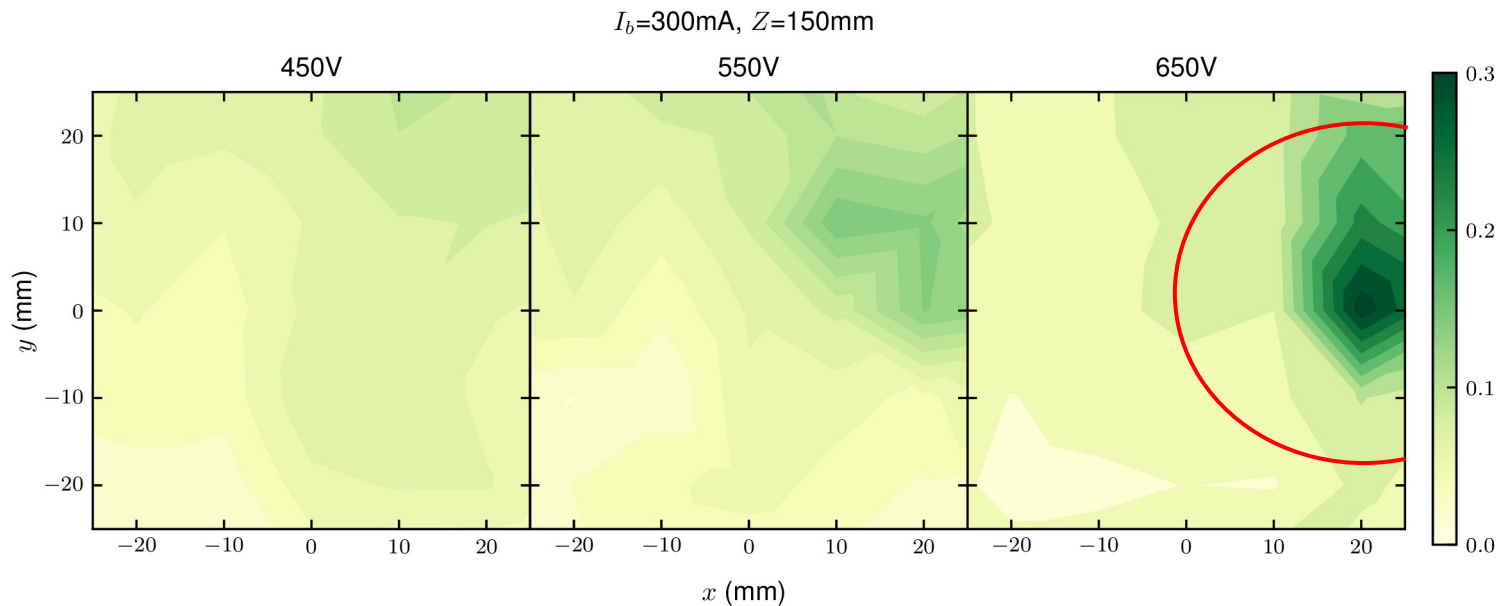


Fig 7. 2D distribution of measured current in A by the RFEA.

- Position system to set the RFEA in a *XYZ space*.
- Study for $I_b = 200, 300\text{mA}$ and V_{acc} between 450 and 650V.
- Axial positions of 150, 250 and 300mm.
- Unfortunately, issues with alignment between thruster and RFEA makes further positions not usable

- Centering data at the maximum current and interpolation of data at different radial directions to obtain *average and deviation* of current and mean velocities for high and low energy populations.
- This allows to compare relevant values at different V_{acc} and I_b .

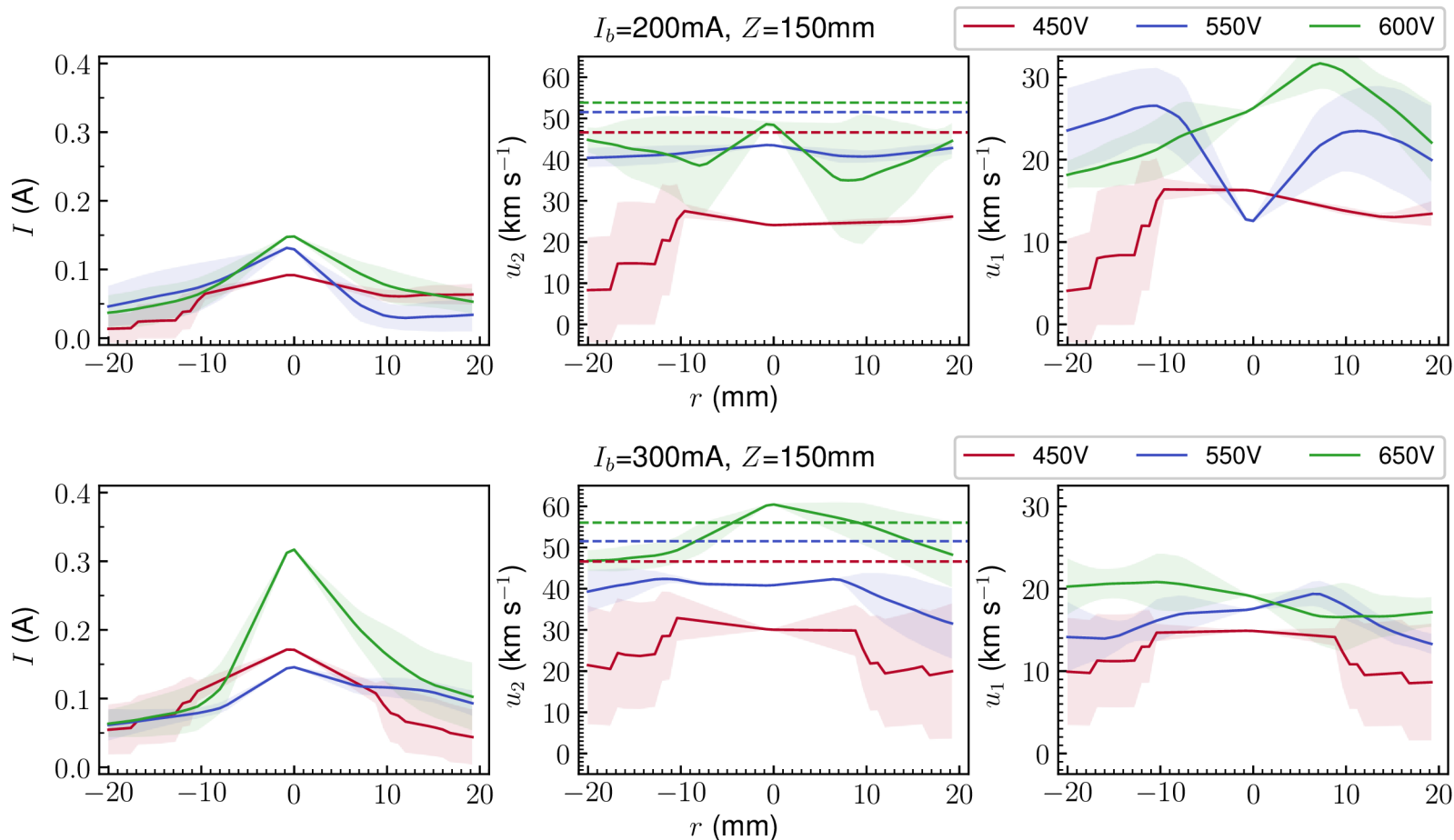
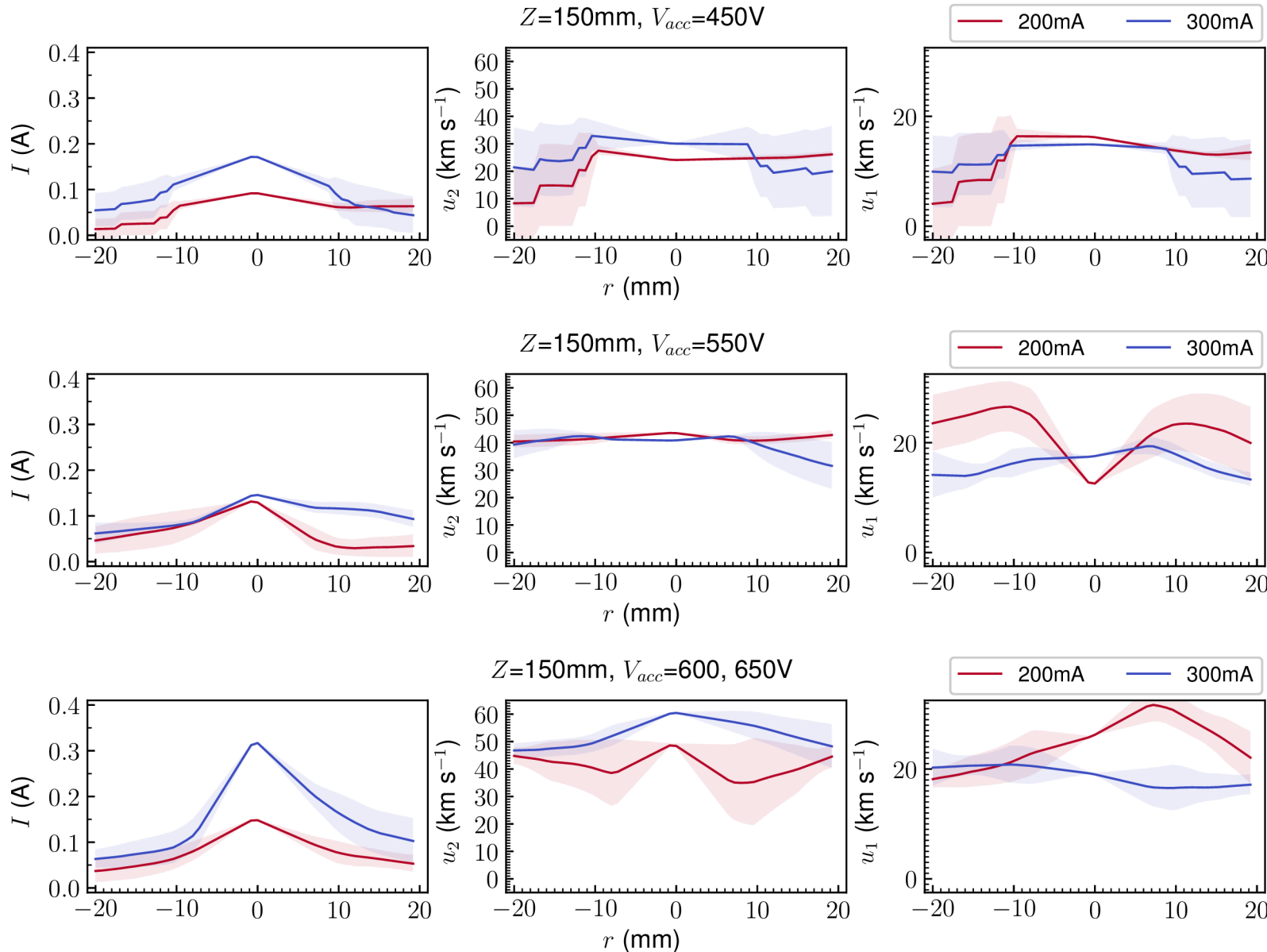


Fig 8. Average values of current, peak velocity of high and low energy populations.

- Increase of acceleration voltage results in an increased ion current.
- Better electrical transparency but also *additional energy* for the inflow of electrons.
- Case of $I_b = 200\text{mA}$ is limited by the acceleration voltage => *not enough influx of electrons*.
- Cases at 450V are always low performance => not enough energy for the influx of electrons.
- *Current is focused* around the $\pm 10\text{mm}$ radius from the point of maximum current => good collimation.

I_b parameter scan



- The increase of available electrons (from current I_b) results in an *increase of ion current without losing axial velocity*.
- This effect is noticeable at all acceleration voltages.
- The case at higher voltage is not the same for the two currents, so that explains the change in velocity.

Fig 9. Average values of current, peak velocity of high and low energy populations.

•Patents:

- 1) L. Conde, J.L. Domenech-Garret, J.M. Donoso, E. Del Río and M.A. Castillo. *Plasma accelerator with modulated thrust*. US patent 10172227B2 (2019). <https://patents.google.com/patent/US10172227B2/en/>
- 2) L. Conde, J.L. Domenech-Garret, J.M. Donoso, E. Del Río and M.A. Castillo. *Plasma accelerator with modulated thrust and space born vehicle with the same*. European Patent EP3369294B1 (2015). <https://patents.google.com/patent/WO2017071739A1/ar/>

•Alphie laboratory testing:

- 3) L. Conde, P.E. Maldonado, J. Damba, J. Gonzalez J.L. Domenech-Garret, J.M. Donoso and M.A. Castillo. *Physics of the high specific impulse alternative low power hybrid ion engine (alphie): Direct thrust measurements and plasma plume kinetics*. J. Appl. Phys. **131**, 023302 (2022). DOI: <https://doi.org/10.1063/5.0067214>
- 4) L. Conde, M.D. Lahoz, J. Grabulosa, R. Hernández, J. Delgado and J. Damba. *Thrust stand base on a single point load cell for impulse measurements from plasma thrusters*. Rev. Sci. Instrum. **91**, 023308 (2020). DOI: <https://doi.org/10.1063/1.5127189>
- 5) L. Conde, J.L. Domenech-Garret, J.M. Donoso, J. Damba, S.P. Tierno, E. Alamillo-Gamboa and M.A. Castillo. *Supersonic plasma beams with controlled speed generated by the alternative low power hybrid ion engine (ALPHIE) for space propulsion*. Phys. Plasmas. **24**, 123514 (2017). DOI: <https://doi.org/10.1063/1.5005881>
- 6) L. Conde et al. *Physics and performance of the Alternative Low Power Hybrid Ion Engine (ALPHIE) for space propulsion*. IEPC-2019-A-643 paper. Proceedings of the 36th International Electric Propulsion Conference. Vienna, Austria. September 15-20 (2019). http://oa.upm.es/58699/1/INVE_MEM_2019_312763.pdf
- 7) J. Damba, P. Argente, P.E. Maldonado, A.Cervone, J.L. Domenech-Garret and L. Conde. *Multiprobe characterization of plasma flows for space propulsion*. J. Phys. Conf. Ser. **958**, 012002 (2018). <https://iopscience.iop.org/article/10.1088/1742-6596/958/1/012002>

•Numerical simulations:

- 8) J. Gonzalez and L. Conde. *Particle-in-cell simulations of ion extraction and acceleration processes in the alternative low power hybrid ion engine (ALPHIE)*. Phys. Plasmas. **26**, 043505 (2019). DOI: <https://doi.org/10.1063/1.5084242>
- 9) D. Dyubo, J. Gonzalez, O. Tsybin and L. Conde. *Charge transport characterization of the alternative low power hybrid ion engine (alphie) with particle-in-cell simulations*. Phys. Plasmas. **28**, 103509 (2021). DOI: <https://doi.org/10.1063/5.0060260>

- *Alphie* is a new disruptive engine technology characterized for a counterflow of charges through its two-grid system.
- Mostly operated with Ar, although Xe and Kr could be employed.
- High specific impulses with low power consumption.
- Measurements along the plume show good collimation and response to V_{acc} and I_b .
- The inflow of electrons require enough energy (high V_{acc}) and flow (high I_b) to provide a high velocity ion population.
- New campaign of plume measurements to improve alignment and measurements resolution.
- Mini version, aimed for cubesats, in the planing.

Acknowledgements

Grant RT2018-094409-B-100 funded by AEI <https://doi.org/10.13039/501100011033> and also by “ERDF A way of making Europe”, by the European Union.

Thank you for your time and interest